

**Raytheon**

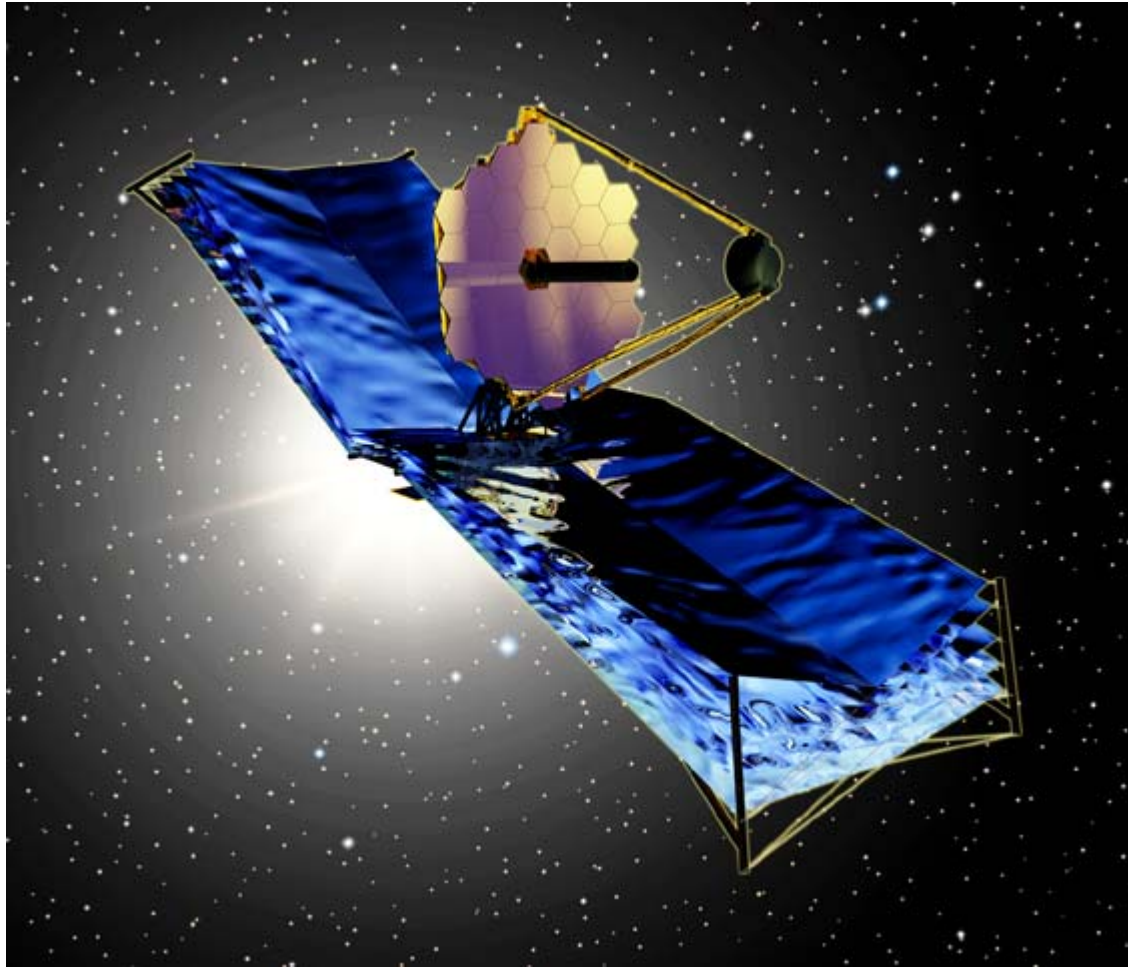
# **Thermal Analysis of Next Generation Space Telescope (NGST) Mirrors During Optical Testing in the X-Ray Calibration Facility (XRCF)**

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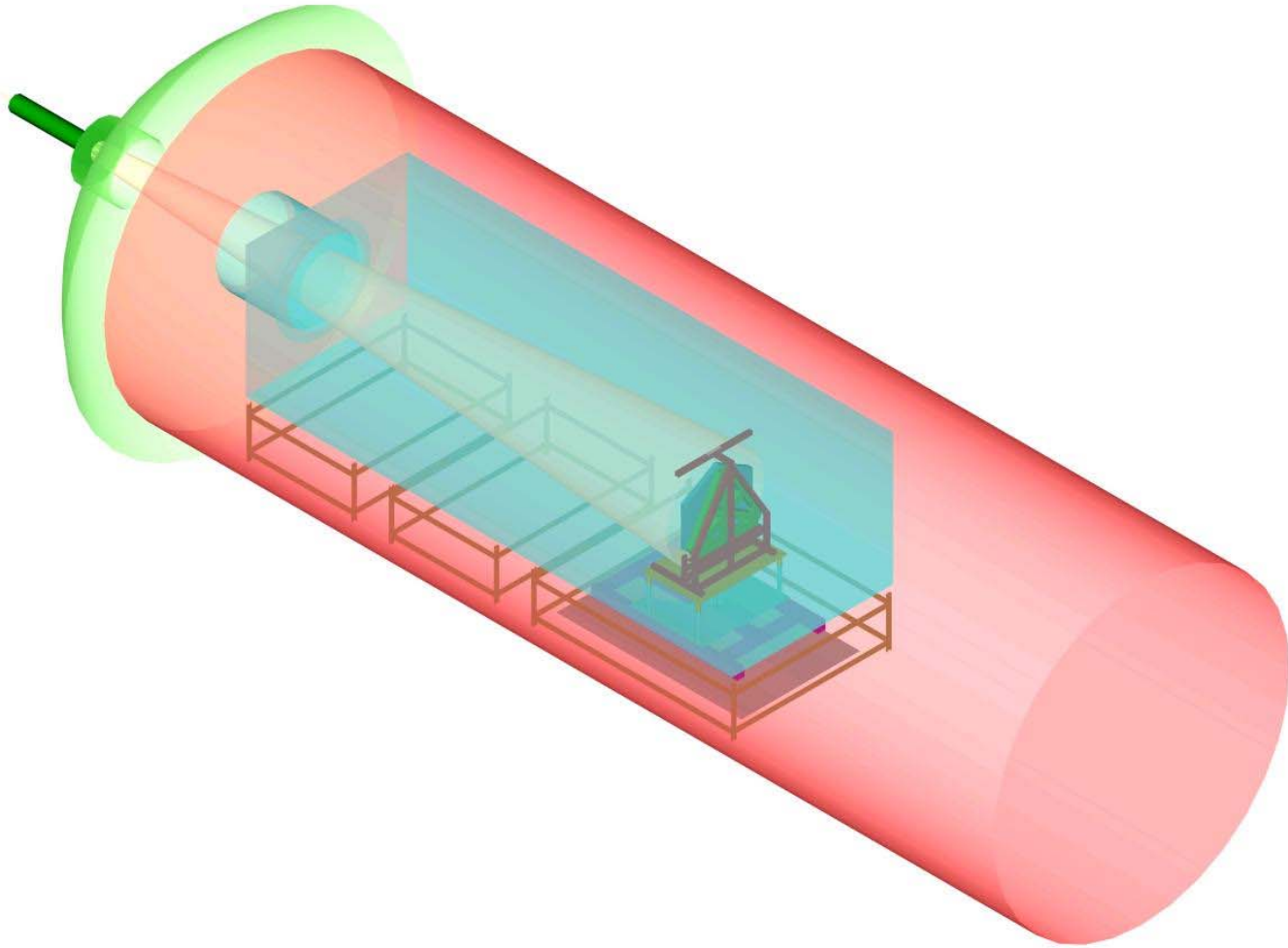
# NGST Spacecraft Concept



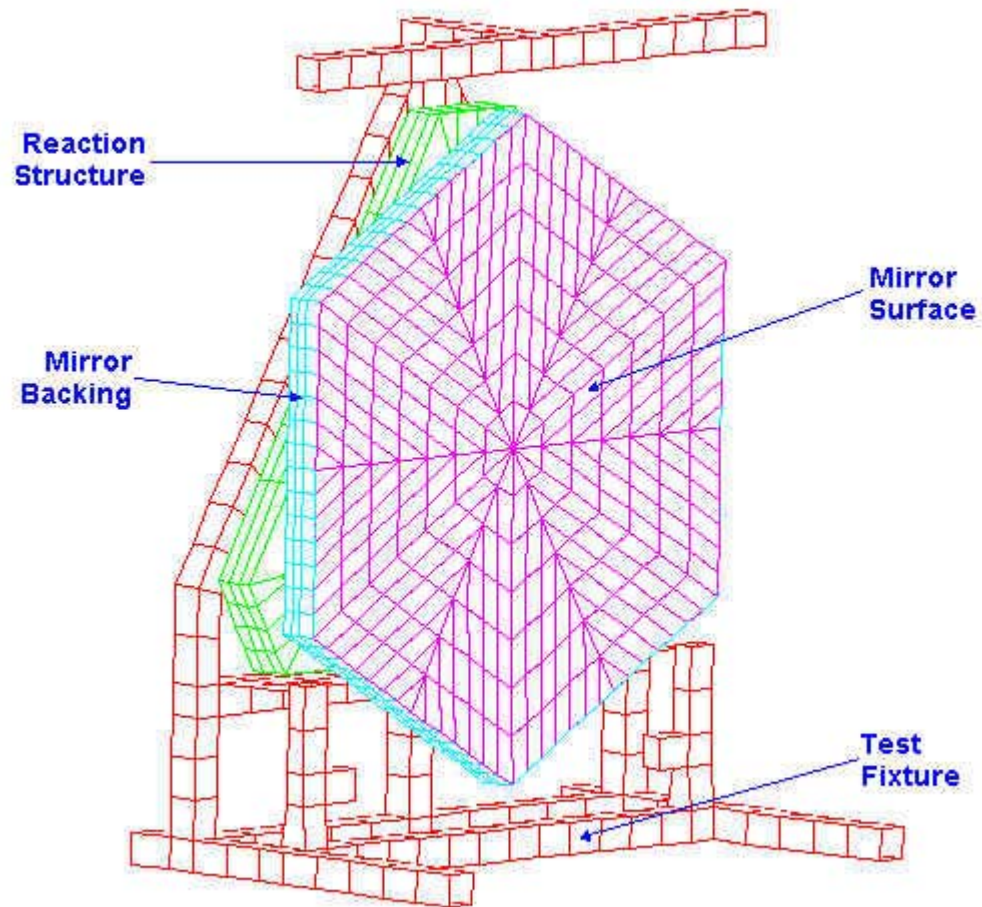
# NGST Mirror Development Testing

- The NGST program and industry partners are developing extremely light-weight mirror designs.
- NGST development mirrors are being tested at MSFC.
- Target temperature for development mirror testing is 35 K.
- Conduction and radiation are not sufficient to conduct thermal vacuum testing in a reasonable time.
- Helium gas is injected into the vacuum chamber to accelerate temperature transitions during testing.
- Free-molecular conduction can be modeled by adapting present thermal analysis techniques.

# NGST Mirror in the XRCF



# NGST Test Article



# NGST Development Mirror



# Knudsen Number

Knudsen Number

$$\text{Kn} = \lambda / L_e$$

Ratio of Mean Free Path to  
Characteristic Dimension

Mean Free Path

$$\lambda = \frac{\mu}{p} \sqrt{\frac{\pi R_u T}{2 g_c M}}$$

$\mu$  = Gas Viscosity  
 $p$  = Gas Absolute Pressure  
 $R_u$  = Universal Gas Constant  
 $T$  = Gas Absolute Temperature  
 $M$  = Gas Molecular Weight

Characteristic Dimension

$$L_e = 4V/A_w$$

$V$  = Enclosure Volume  
 $A_w$  = Enclosure Surface Area

Heat transfer and flow regimes are defined in terms of the Knudsen number.

Continuum                       $\text{Kn} < 0.01$

Mixed                         $0.01 < \text{Kn} < 0.30$

Free-molecular               $\text{Kn} > 0.30$

# Free-Molecular Conduction

The radiation heat transfer between two gray surfaces may be represented by:

$$Q = \sigma F_e F_{12} A_1 (T_2^4 - T_1^4).$$

Similarly, the free-molecular conduction between the two surfaces may be represented by:

$$Q = G p F_a F_{12} A_1 (T_2 - T_1), \text{ in which}$$

$$G = \frac{\gamma + 1}{\gamma - 1} \sqrt{\frac{g_c R_u}{8 \pi M T}}$$

$p$  = Gas Absolute Pressure and

$F_a$  = Accommodation Coefficient Factor.

$R_u$  = Universal Gas Constant

$M$  = Gas Molecular Weight

$T$  = Pressure Gauge Absolute Temperature



# Accommodation Coefficient

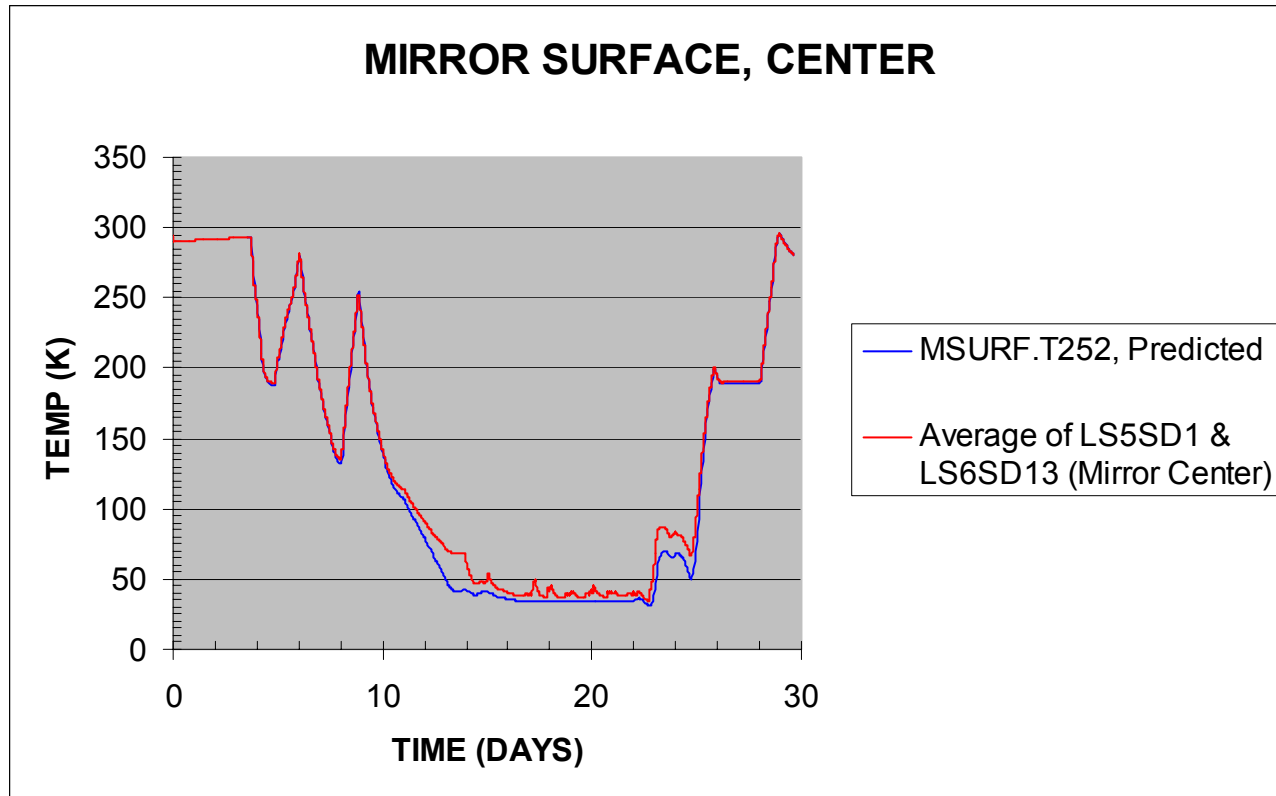
The Accommodation Coefficient,  $\mathbf{a}$ , is analogous to the Emissivity,  $\mathbf{\epsilon}$ , and its value depends on the specific gas/surface combination and the surface temperature. It represents the degree of approach of the gas molecules to thermal equilibrium with the bounding surfaces.

## Accommodation Coefficients for Gases of Interest

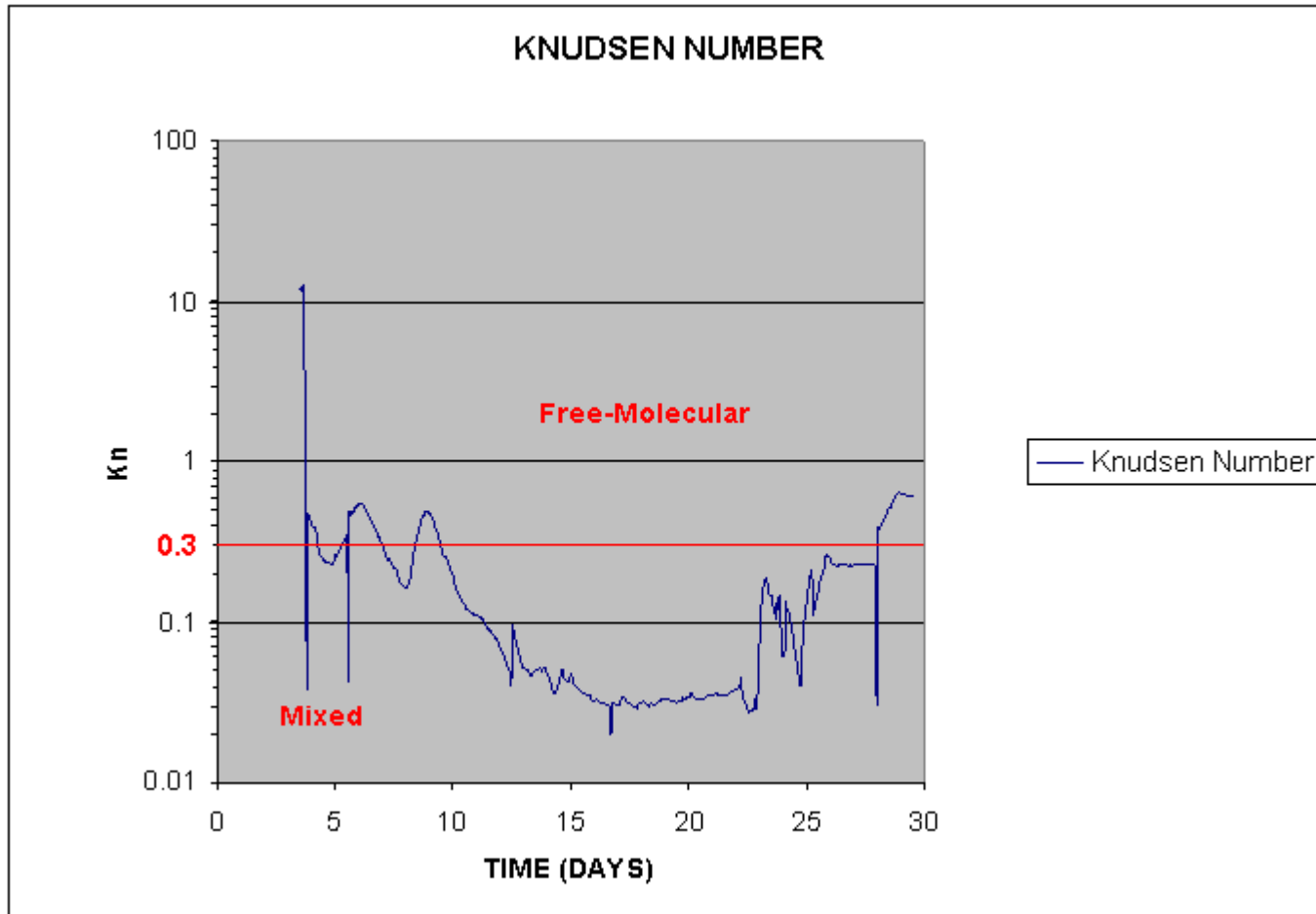
Temp. (K)	He	Air	Ne	H <sub>2</sub>
20	0.59	1.00	1.00	0.97
78	0.42	1.00	0.83	0.53
300	0.29	0.8-0.9	0.66	0.29

$F_a$ , the Accommodation Coefficient Factor, is analogous to the Emissivity Factor,  $F_e$ . All expressions for  $F_e$  may be applied directly by substituting  $\mathbf{a}$  for  $\mathbf{\epsilon}$ .

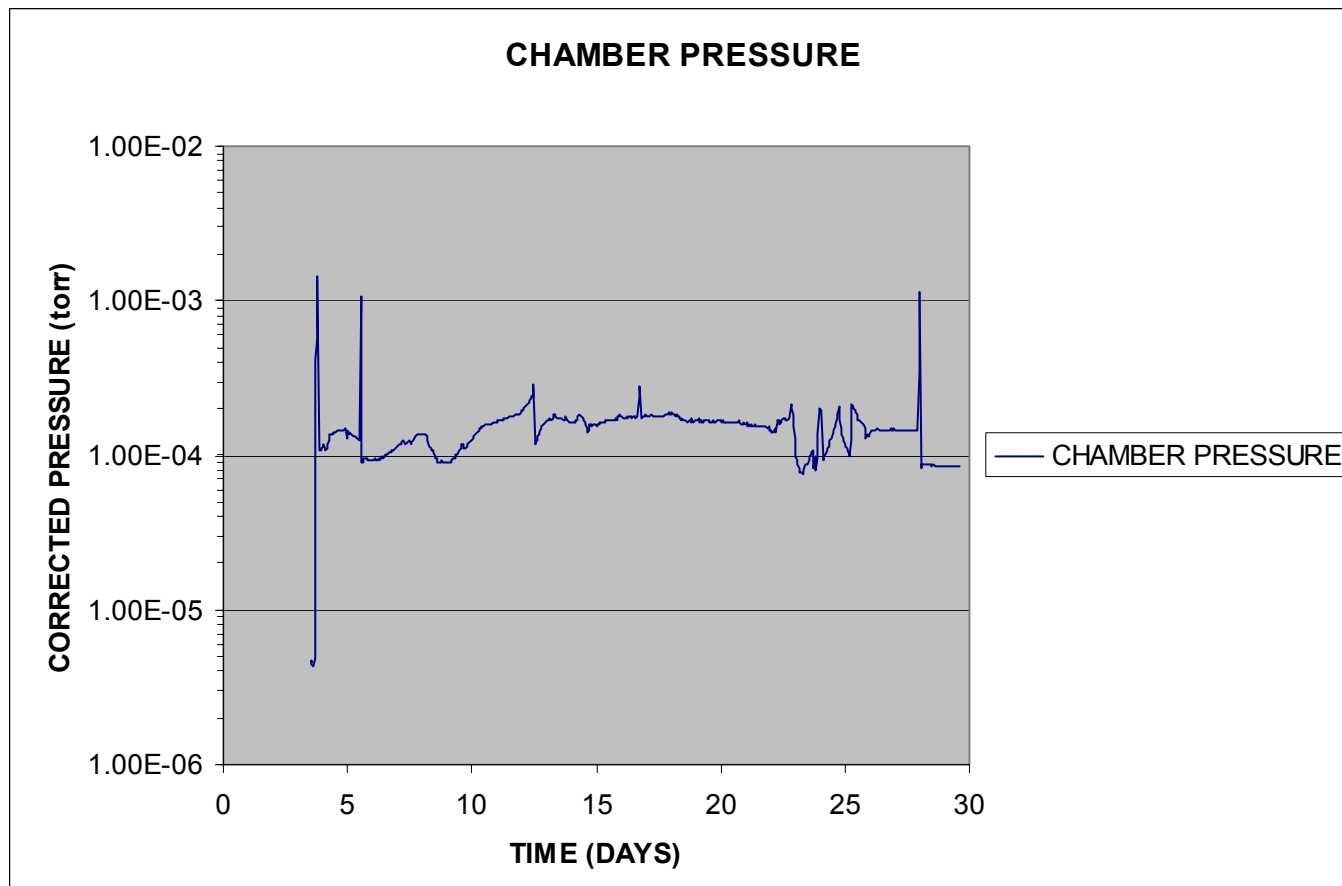
# Results



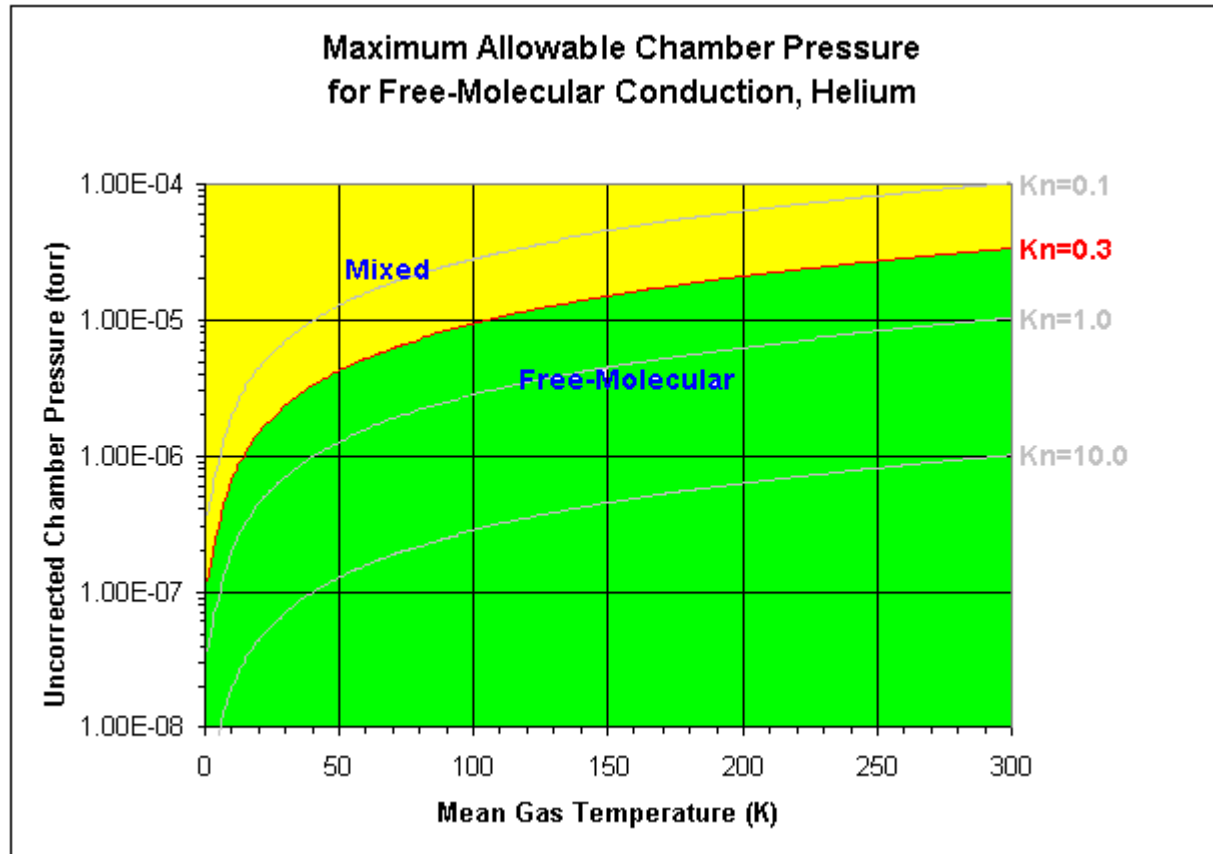
# Results



# Results



# Recommendations



## References

1. Randall F. Barron, *Cryogenic Heat Transfer*, Taylor & Francis, 1999, pp. 243 – 257.
2. Steven Sutherlin, *The Composite Optics, Inc. (COI) Development Mirror Thermal Analysis Final Report*, MG-01-498, May 30, 2001.